

# Agenda

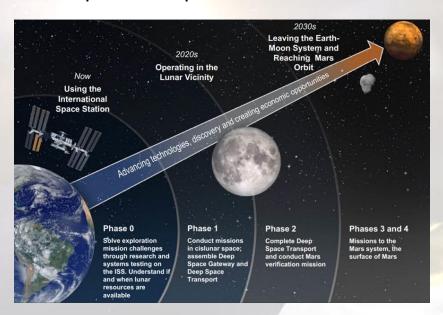


- History of EVA Technology Development efforts
- xEMU Lite Flight Demonstration Hardware Development Plan





- Since 2007, there has been a continuous effort by NASA to develop EVA technologies to enable future exploration missions
  - Incorporating lessons learned from 30+ years of EMU operations
  - Designing for the different environments of the potential destinations
  - Developing hardware that enables scientific exploration and supports the operational concepts of the potential destinations



 Development has occurred under several NASA programs but the team and overall development plan have remained essentially the same



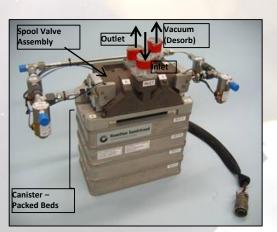
- 2007-2010: Exploration Technology Development Program (ETDP) focused on technologies for a lunar mission
  - Performed a PLSS schematic trade study to determine the combination of life support technologies that would best meet exploration mission needs
  - Advanced new system-enabling PLSS component technologies such as the Suit Water Membrane Evaporator (SWME) and Rapid Cycle Amine (RCA)
  - Began work on lunar-focused PGS technologies including bearings,
     Thermal Micrometeoroid Garment (TMG), and gloves



**SWME 2.0** 



RCA 1.0



Primary Gox

Recharge

Recharge

Shut Off Valve

SWME

Resistive Heater

To Ambient

Early Exploration PLSS Schematic



- 2010-2011: Enabling Technology Development and Demonstration (ETDD) had the goal of developing and demonstrating prototype systems to support exploration goals
  - Continued EVA efforts from ETDP with the addition of SE&I activities
  - Integrated components into systems
  - Culminated in building and testing PLSS 1.0 and Z-1





Z-1 Prototype Suit



- 2011-2016: Advanced Exploration Systems (AES) and Space Technology Mission Directorate (STMD)
  - Focused on maturing component technologies, integrating them into prototype systems, and demonstrating them in testing
  - Culminated in building and testing PLSS 2.0 and Z-2





PLSS 2.0 Human-in-the-Loop Testing



**Z-2** Prototype Suit



#### 2016-2017: ISS Program

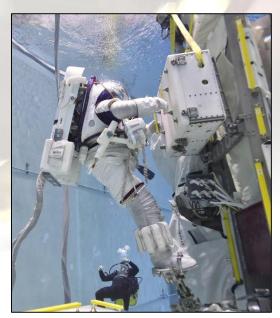
- Continued to advance the TRL of EVA Suit technologies with the goal of developing a NASA reference concept with the capability to support missions at ISS and cis-lunar space
- Focused near-term development on a "Lite" version of the xEMU, which defers some capabilities
- Assembled and conducted an electrical live loads test of the xPLSS
- Performed 19 NBL runs with Z-2



xPLSS Electrical Live Loads Configuration



Z-2 Ingressing the Airlock in the NBL



Z-2 Foot Restraint Evaluation in the NBL



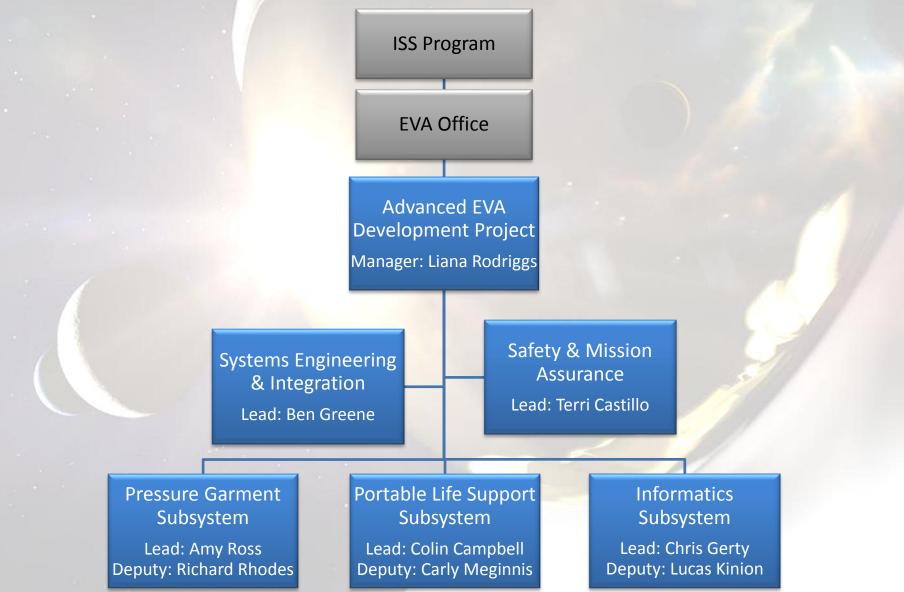
## Transition to Flight Hardware Development



- The technologies developed over the last 10 years are now being pulled into a flight hardware development effort
- Starting in FY18, the project objective is the development of the xEMU Lite for a flight demonstration on ISS
  - Demonstrate core suit capabilities needed for the full xEMU for exploration missions
- The in-house NASA Advanced EVA Development team that has been performing EVA technology development will design the xEMU Lite and build a single flight demonstration unit
  - NASA will be procuring components and will perform the role of system integrator

# **Project Organization**



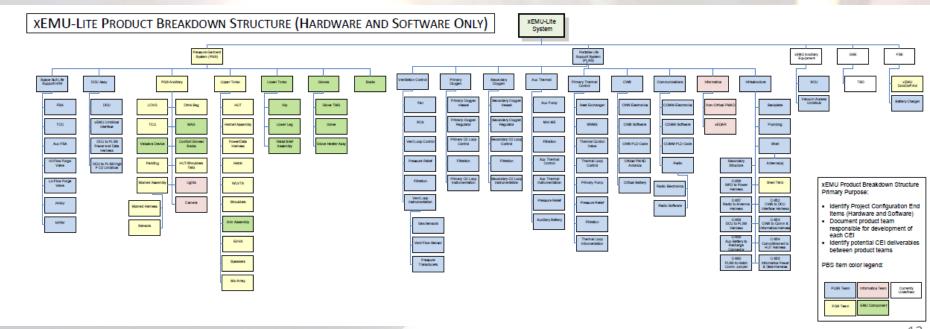


### **xEMU Lite Architecture**



- The xEMU Lite is a subset of the fully outfitted xEMU, with non-critical capabilities deferred
- Details of the architecture are captured in the Architecture Description Document
- The Product Breakdown Structure identifies and organizes the hardware and software deliverables for the project

System	Lite-Configuration		
PLSS	xPLSS Lite		
PGS	xPGS Lite, EMU Softgoods		
INFO	xINFO Lite		
FSE	New ISS FSE		
T&E	ISS Standard Tools		



# xEMU Lite vs xEMU





# **xEMU Lite**ISS Demonstration

	xEMU Lite	Feature	xEMU
1	4.3 psi	Operating Pressure	8.2 psi
	LEO	Design Environment	Deep Space
	Microgravity		Microgravity
			Surface
	Upper Torso + Min. Lower Torso	Mobility	Upper Torso + Full Lower Torso
	Scarred for future upgrade	Crew Autonomy	Graphical Display

**xEMU**Deep Space EVA
For
Gateway and Mars Transit

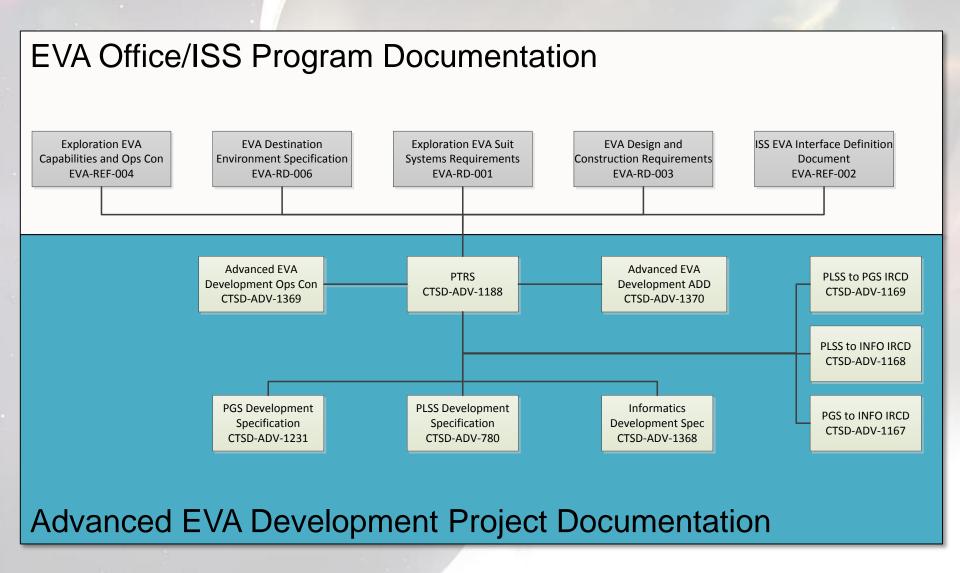
### **xEMU Lite Development Assumptions**



- 3 complete hardware iterations will be built
  - Design Verification Test (DVT) prototype pre-qualification testing
  - Qualification unit
  - Flight unit
- The flight demo will be performed with a single xEMU Lite suit
- Flight demonstration objectives and a concept of operations will be developed prior to SRR
  - Dual suit operations with EMU
  - Multiple EVA's over a period of time
- Goal is to design the xEMU Lite to meet as many of the xEMU requirements as feasible within cost and schedule constraints, but it will be certified for a demo, not for full EMU replacement or xEMU (ex. Life requirements)
  - Details are being worked out in preparation for the project Systems Requirement Review (SRR)
- Work is underway to determine best approach to interface xEMU Lite with the ISS airlock
  - Permanent vs. temporary approaches are being considered

# Requirements Structure/Doc Tree





### Tentative Schedule



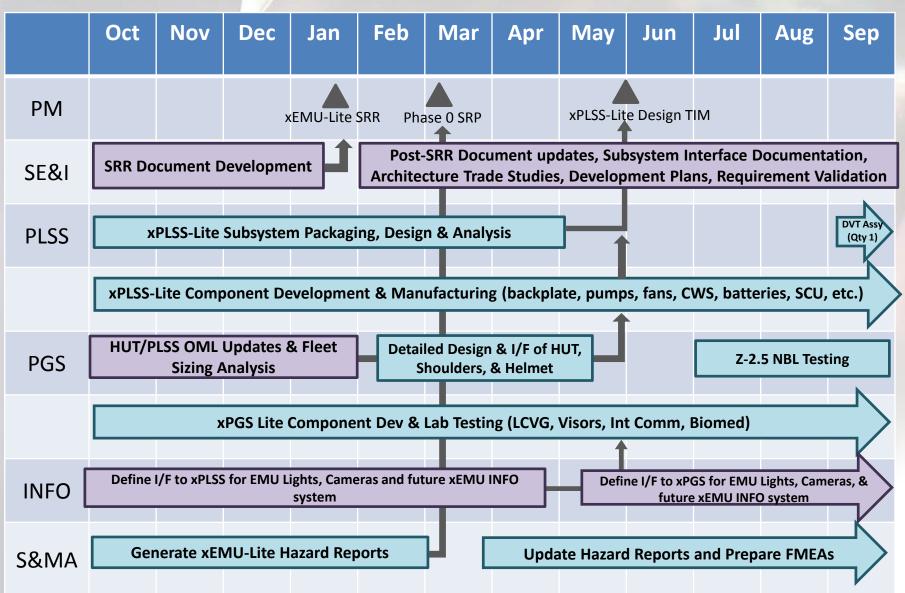
	FY18	FY19	FY20	FY21
xEMU Lite Milestones	SRR	PDR	DVT Build/Test	CDR

Terms and Definitions: SRR – System Requirements Review, PDR – Preliminary Design Review, CDR – Critical Design Review, DVT – Design Verification Testing

- Project-level System Requirements Review (SRR) in January 2018
- PLSS Subsystem design TIM in late spring 2018
  - Informal peer review
- PGS Subsystem design TIM in fall of 2018
  - Informal peer review
- Project-level Preliminary Design Review (PDR) in mid-2019
  - Initial assumption is that we will have a series of component PDR's leading to the system review
- Project CDR in FY21
- Flight demonstration by mid-2020's

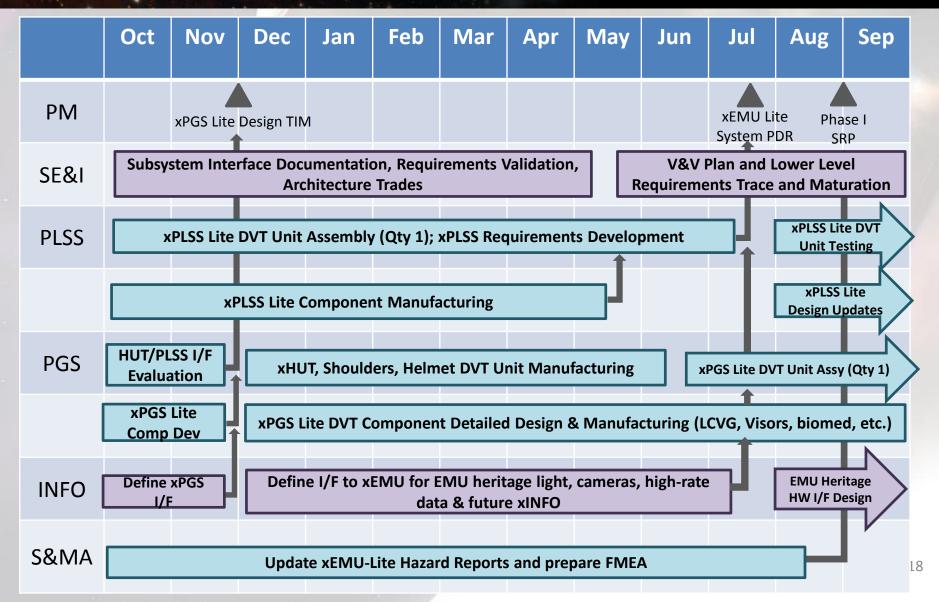
### FY18 Plan Overview





#### FY19 Plan Overview





<sup>■</sup> Hardware development ■ Requirements & I/F & Testing

